

The Direct Strength Method

In Cold-Formed Steel Design





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About ClearCalcs.com

ClearCalcs helps engineers design without compromise by bringing together powerful FEA analysis with easy to use design tools for concrete, steel, cold-formed steel and timber.

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Intro Video Hyperlink



More Accurate

Design more accurately with unrestricted and accessible FEA analysis



Eliminates Wasted Time Eliminate time wasted using clunky methods or waiting for software licenses to free up



Available Everywhere Empower engineers to work effectively from office, home, or site

Meet the Presenter

- Brooks H. Smith | Head of Engineering R&D
 - Chartered Professional Engineer (AU) & P.E. (USA)
 - MCivE from University of Massachusetts
 - BEng from Dartmouth College
 - 8 years of previous experience in:
 - Structural engineering R&D consulting, specialising in cold-formed steel
 - Research fellowship in system behaviour of thin-walled steel
 - Forensic structural engineering, specialising in reinforced and PT concrete
 - Almost 5 years now with ClearCalcs
 - Focusing on R&D and QA



ClearCalcs

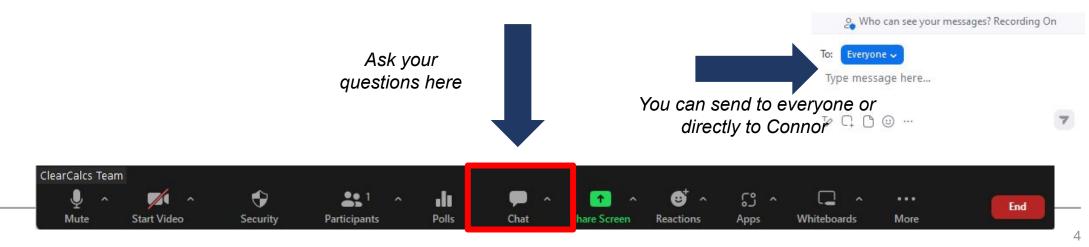


How to Ask Questions

Meeting Chat

• Type your questions in the Chat tab on your Zoom control panel and click Send

- Please send your questions to everyone
- We will address all questions in the second half of the webinar during the 15-minute Q&A session
- We might invite you to unmute yourself to ask your question live!





Agenda – Today's Goals

Theory on the Direct Strength Method

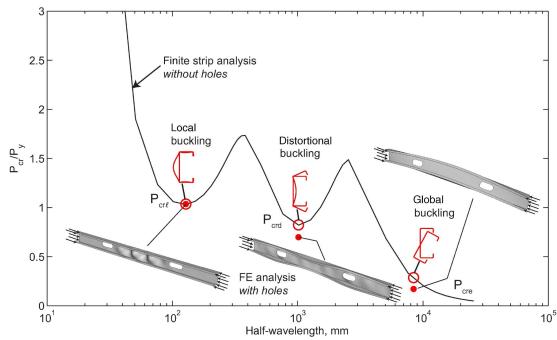
- DSM vs Effective Width Method
- Finite Strip Method

• Practically Using DSM

- Keeping Finite Strip Method Simple
- Using DSM in AS/NZS 4600:2018

Demonstration in ClearCalcs

- Just the Design
- Doing a Finite Strip Method Analysis



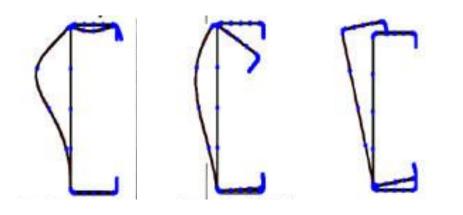


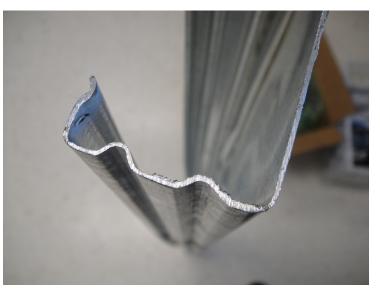
Theory on the Direct Strength Method

Comparing with the Effective Width Method, and doing a finite strip analysis

Buckling in Cold-Formed Steel

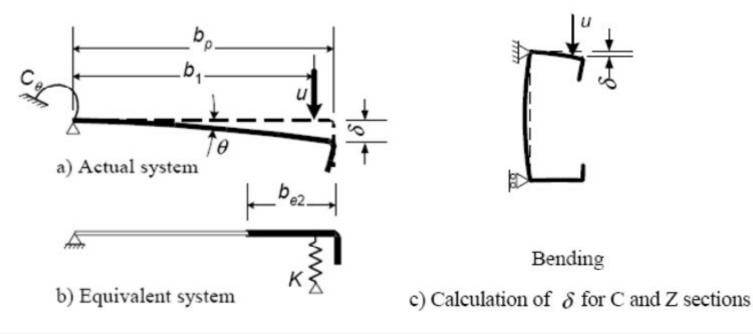
- Hot-rolled steel classifies sections as compact, non-compact, or slender and requires extra equations for "slender"
 - In cold-formed steel, "slender" checks basically always need to be done
- Local, distortional, or global buckling for bending or compression
 Global encompasses both lateral and torsional buckling
- Stiffeners function to mitigate buckling





Old Method - Effective Width

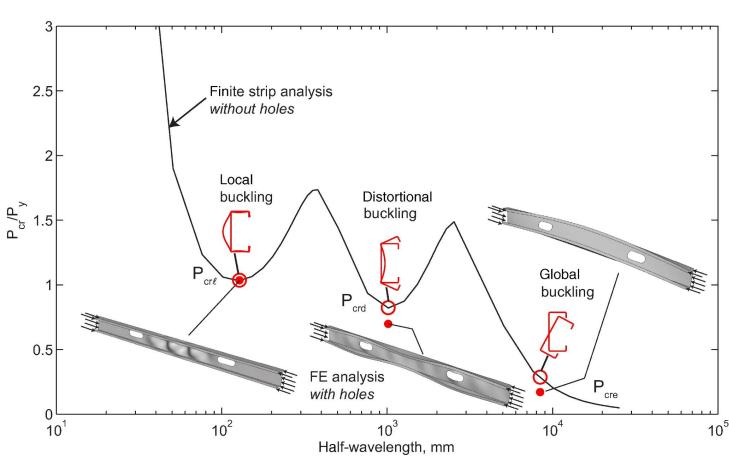
- At its most basic, accounts for buckling by pretending that each 'plate' within an element is shorter than it really is
 - More susceptibility to buckling = shorter 'effective width'
- Separate equations for local vs distortional vs global buckling





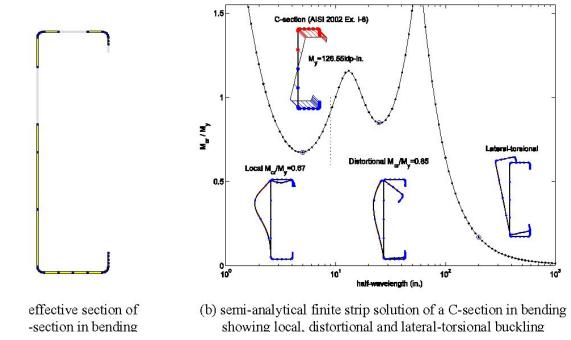
New Method - Direct Strength

- Directly consider the strength and stiffness of the cross-section as a whole
- Consider every half-wavelength for buckling susceptibility
- Every type of buckling in one unified method



EWM vs DSM: Why Switch?

- More flexible: EWM is a PIA for complex sections/many plates
 - But every section is exactly as much work for you in DSM!
- More accurate: DSM is will give you up to ~10% more capacity
- Easier calculations: Maths mostly done by finite strip analysis





Interjection: Buckling Modes

- Local Buckling: corners of the cross-section stay still, while the flat plates bend
 - Usually occurs at a half-wavelength of about 100-250 mm
- **Distortional Buckling:** corners of the cross-section move, but not all corners move together
 - Usually occurs at a half-wavelength of about 400-800 mm
- **Global Buckling:** whole cross-section rotates or translates as a single unit
 - Usually occurs at a half-wavelength greater than ~1.5 m

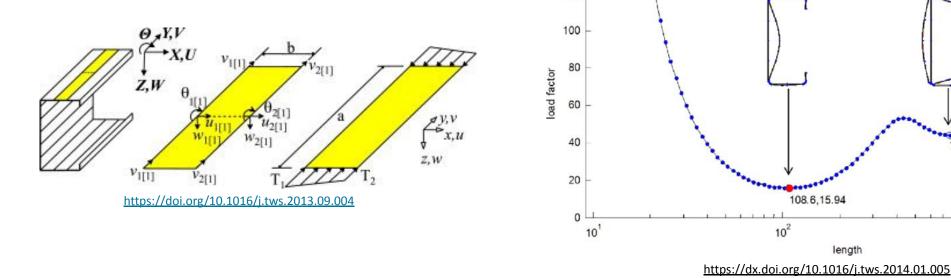
ClearCalcs.com | FEA Structural Design in the Cloud

What is the Finite Strip Method?

- The Direct Strength Method requires a rational analysis that usually takes the form of the Finite Strip Method
 - Instead of finite elements of little triangles or rectangles, we have entire *strips* of arbitrary length

120

• All buckling modes



(6) ClearCalcs

760.5.44.

 10^{3}

108.6.15.94

length

 10^{2}



Finite Strip Software

- A couple main pieces of software available:
 - **<u>CUFSM</u>** (free), from Johns Hopkins University
 - **<u>pyCUFSM</u>** (free), ported to Python by Brooks Smith
 - **THIN-WALL** (paid), from the University of Sydney





Centre for Advanced Structural Engineering THIN-WALL



Finite Strip - Signature Curves

- Commentary to the **US's AISI \$100** standard is very helpful (and free!)
 - AS/NZS 4600's DSM section is almost verbatim copied from **AISI S100**
 - It's just in Imperial instead of Metric units!

Understanding axis is applied to this section. All results are given in reference to this applied stress distribution. Any axial stresses (due to bending, axial load, Finite Strip Analysis Results warping torsional stresses, or any combination thereof) may be considered in the analysis. Mode shapes are shown at C-section (AISI 2002 Ex. I-8) the identified minima and at Minima indicate the 200 in., Identification of the lowest load level at M_=126.55kip-in. mode shapes is critical to which a particular DSM, as each shape uses a mode of buckling different strength curve to occurs. The lowest connect the elastic buckling M_{cr}/M_v is sought for 111 results shown here to the each type of buckactual ultimate strength. In ling. An identified the section. local buckling cross-section mode Lateral-torsional only involves rotation at shape can repeat Distortional M_/M_=0.85 internal folds, distortional along the physical Local M_/M_=0.67 buckling involves both 0.5 rotation and translation of internal fold lines, and lateral-torsional buckling involves "rigid-body" deformation of the crosssection without distortion. 100 10² 101

Applied stress on the section indicates that a moment about the major

half-wavelength

25 in. Distortional

200 in. Lateral-torsional

5 in. Local

Figure 2 Understanding Finite Strip Analysis Results

half-wavelength (in.)

variation along the member length

length of the

Half-wavelength

shows how a given

cross-section mode

shape (as shown in the figure) varies

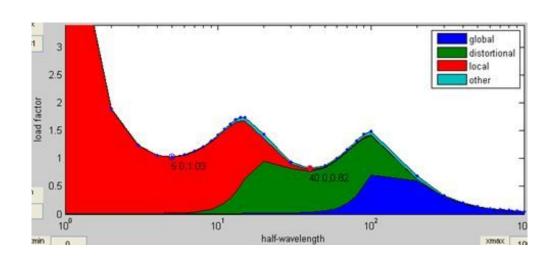
along its length.

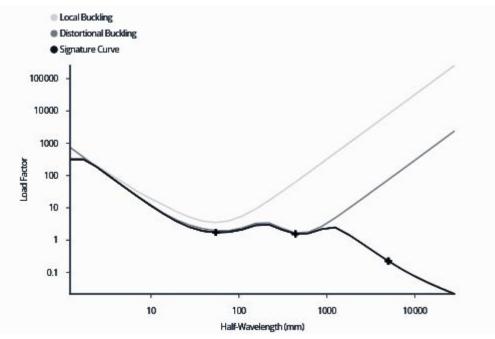
member.



Finite Strip - Mode Classification

- Signature curves are not always as clear as previous example
- "Constrained Finite Strip Method" (cFSM); two methods:
 - 1. Classifying a signature curve, or
 - 2. Multiple FSM analyses, with 'constrained' modes







Practically Using DSM

Finite Strip Analysis in real life, and how it fits with AS/NZS 4600:2018

Direct Strength Method Req'ts (Cl 7.1.2)

- DSM is applicable to most sections you may encounter
 - But should still check this:

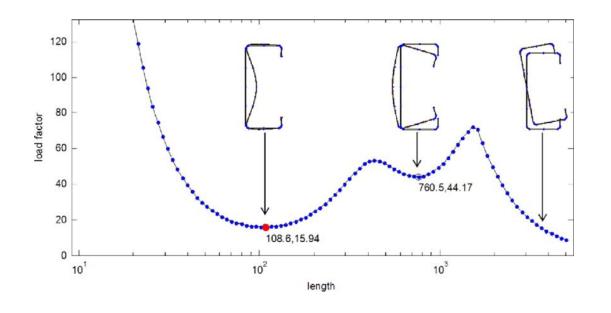


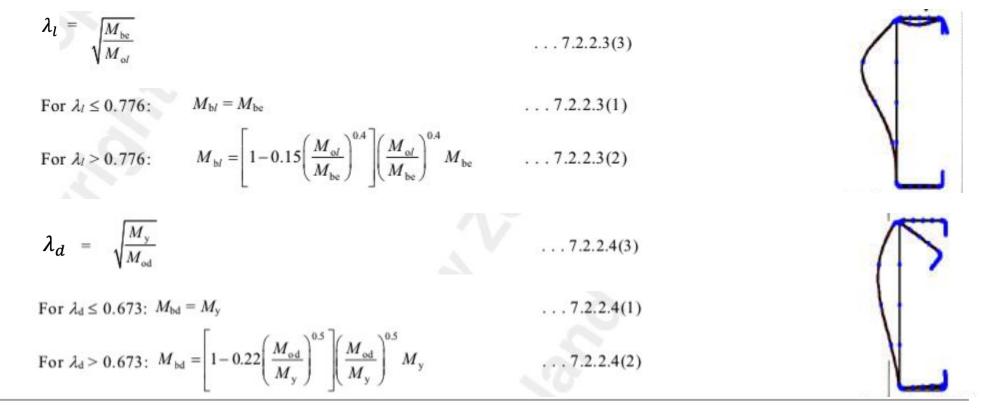
TABLE 7.1

LIMITS OF APPLICABILITY FOR DESIGN USING THE DIRECT STRENGTH METHOD

Criteria	Limiting variables	DSM prequalification limits	
Stiffened element in compression [Figure 1.3(C)]	<i>b</i> ₂ / <i>t</i>	≤500	
Edge stiffened element in compression [Figure 2.4.2(a)]	b/t	≤160	
Unstiffened element in compression [Figure 2.3.1(a)]	d/t	≤60	
Stiffened element in bending [Figure 2.2.3(a)]	b/t	≤200 for unstiffened web ≤260 for bearing stiffener ≤300 for bearing and intermediate stiffener	
Inside bend radius	r_{\min}/t	≤20	
Simple edge stiffener overall length/overall width ratio	$\frac{\left(d+r_{\min}+t\right)}{\left(b+2r_{\min}+2t\right)}$	≤0.7	
Maximum number of intermediate stiffeners in <i>b</i> 2	nf	4	
Maximum number of intermediate stiffeners in <i>b</i>	n _{fe}	2	
Maximum number of intermediate stiffeners in web	n_w	4	
Yield stress used in design	fy	≤655 MPa	

Local & Distortional Buckling

- Just a couple simple equations for determining if that buckling mode will occur, and then calculating the actual buckled capacity
 - Similar equations for bending moment or axial compression



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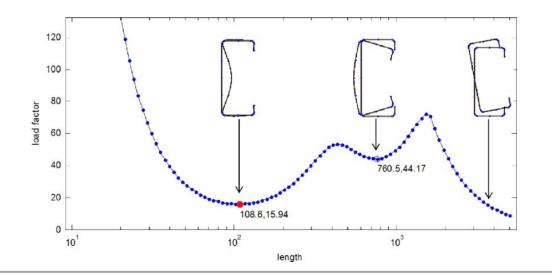
Global Buckling

- Critical buckling factor can be from Finite Strip Method results
- But, for standard sections, analytical formulae can be easier
 - Just need formulae for f_{ov} , f_{oz} , and (for axial) f_{ovz}

 $M_{\rm o} = C_{\rm b} A_{\rm g} r_{\rm ol} \sqrt{f_{\rm oy} f_{\rm oz}}$

... D2.1.1(1)

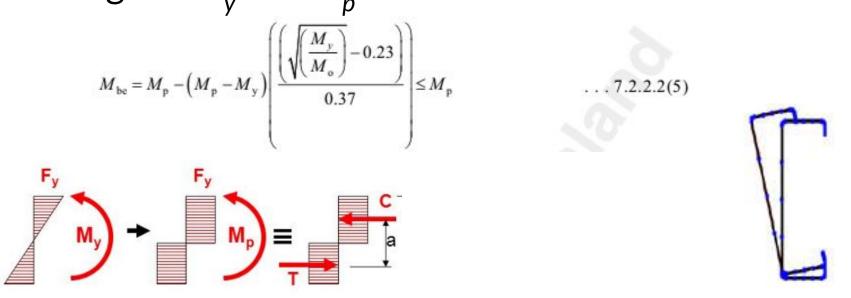
• Then similar equations as local and distortional buckling





Inelastic Reserve Capacity

- Allows small amounts of localised yielding that don't affect stability
 - Optional; certain connections (like welds) may forbid it
 - Only allowed if critical global buckling >> yield strength
- Weighted average of M_v and M_p



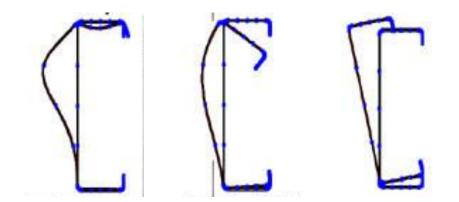


Final Capacity

- Just simply the minimum of local, distortional, and global buckling capacities
 - $\phi_{h} = 0.9$ in bending, or 0.85 in compression

$$\phi_{b}M_{b} = 0.9 * min(M_{bl}, M_{bd}, M_{be})$$

$$\phi_c N_c = 0.85 * min(N_{cl}, N_{cd}, N_{ce})$$





Load Interactions

- **Bending + Shear:** some equations for shear differ slightly
 - We need compatible capacities so that we can combine them

$$\left(rac{M^{*}}{\phi_{
m b}M_{
m s}}
ight)^{2} + \left(rac{V^{*}}{\phi_{
m v}V_{
m v}}
ight)^{2} \leq 1.0$$
 7.2.3(9)

- Note that M_s is a slightly different value it's M_{bl} but without consideration of global buckling
 - i.e. $\lambda_1 = \sqrt{M_v/M_{ol}}$ instead of $\lambda_1 = \sqrt{M_{be}/M_{ol}}$
- **Bending + Axial:** first, separately calculate pure bending and pure axial capacities using the Direct Strength Method

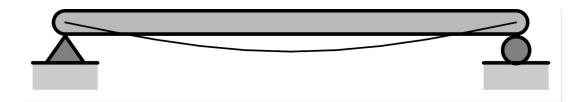
$$rac{N^{*}}{\phi_{
m c}N_{
m c}} + rac{M_{
m x}^{*}}{\phi_{
m b}M_{
m bx}} + rac{M_{
m y}^{*}}{\phi_{
m b}M_{
m by}} \leq 1.0$$
 7.2.4



Deflections

- Also more accurate using our DSM results
 - Conservatively, can use I_{eff} values given by manufacturers
 But more accurate is to use our DSM moment capacities
 - - *M* is the maximum service moment demand
 - $M_n = M_b$, except that we replace all instances of M_v with M

$$I_{\rm eff} = I_{\rm g} \left(\frac{M_{\rm n}}{M} \right) \le I_{\rm g}$$





Demonstration in ClearCalcs

How does the workflow look like in ClearCalcs



Questions?





THANK YOU!

- We will send you a recording of the webinar by email.
- There will be a survey at the end of this webinar, we would appreciate your feedback on how we can improve.
- If you have further questions, send an email to <u>help@clearcalcs.com</u> or use the Help button in ClearCalcs
- Stay tuned for webinar [Webinar Title] next month!



Appendix

About ClearCalcs



Happy Engineers Using ClearCalcs

ClearCalcs has been used in 2,000,000+ designs by a growing number of engineers across the globe, with the US becoming our largest customer base in 2021.







"You are light years ahead of the competition on features and ongoing growth." "Why didn't you just use ClearCalcs for that?"

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Don C. Foundation Engineering Specialists, LLC

Helen W. via Landon R. Criterium Engineers **Jason M.** J. Michael Engineering, PLLC





The ClearCalcs Team

A growing team of passionate engineers, programmers, customer success specialists, product managers, marketers, and more!





What Sets Our Calculations Apart

• Live solutions

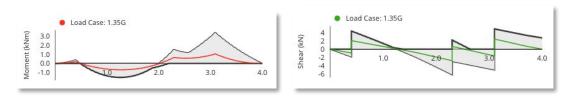
• Instantly see how every change you make affects the design, in all load cases

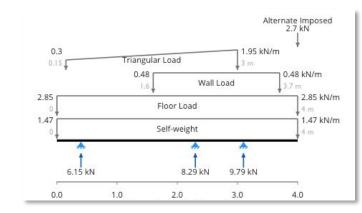
• Finite Element Analysis

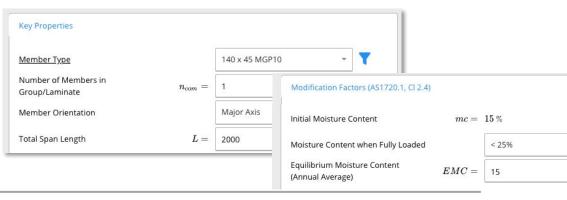
• Get the most accurate results no matter what your configuration

• As simple or complex as you want

• Safely enter in only a few properties, or tune every parameter – it's up to you







What Sets Our Design Process Apart

Member selector

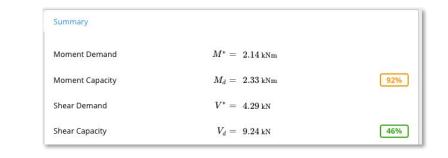
- Check every possible member in seconds
- Link your loads
 - No need to manually copy reactions into the next sheet just create a link

• Simple traffic light indicators

• See at a glance how close your design is to perfection

Designation	M_d V_d δ_l δ_s
70 x 35 F5 Seasoned SW	450% 91% 417% 752%
90 x 35 F5 Seasoned SW	273% 71% 198% 354%
120 x 35 F5 Seasoned SW	154% 53% 84% 150%
140 x 35 F5 Seasoned SW	113% 46% 53% 95%
190 x 35 F5 Seasoned SW	62% 34% 22% 38%

Roof	Lintel RL8	1			÷
Support	Location G (mm)	overning Reactions F R* (kN)	ermanent Load Reactions R*_G (kN)	Imposed Load Reactions R*_Q (kN)	
Support		overning Reactions F R* (kN) 0.293	Permanent Load Reactions R*_G (kN) 0.0667	Imposed Load Reactions R*_Q (kN) 0.133	





What Sets Our Platform Apart

• Clean, clear printouts

Beautiful results your clients can underst

See full detail for every field

• References, equations, and more

• Rapid product updates

• Receive new features and calculations within days, not years

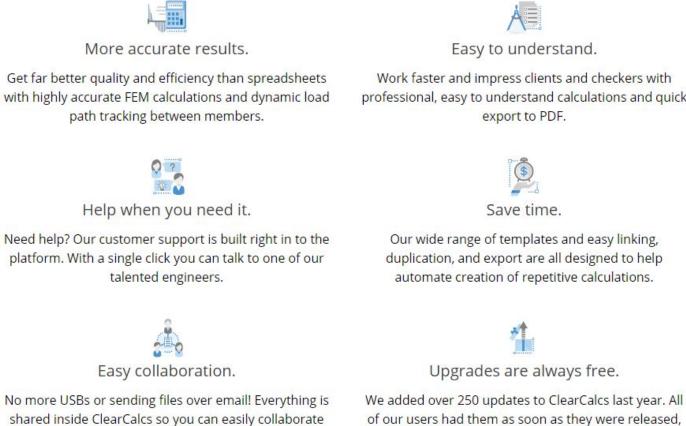
	I	ல	Client: Engineer: Brooks Smith Project: test	Date: Oct 17, 2018 Job #: Subject: 87		
erstand		Moment Demand 85% Moment Capacity Shear Demand 18% Shear Capacity	about X-Axis $M_x^* = 10.3$ for	Nm Nm + M _{ent} N		
ersturiu		Shear Capacity (AS4100-1998, Section 5.11) Shear Capacity Factor $\phi = 0.9$ Nominal Shear Yield Capacity $V_w = 131 \text{ km}$ Nominal Shear Buckling Capacity $V_b = 131 \text{ km}$ Nominal Shear Capacity in Uniform $V_w = 131 \text{ km}$ Stress Distribution $V_w = 131 \text{ km}$ Nominal Shear Capacity $V_w = 131 \text{ km}$ Stress Distribution $V_w = 131 \text{ km}$				
Weak Axis Buckling Stress	$f_{oy} = ~112 \mathrm{~MPa}$		Moment Section Capacity (AS4100-	1998, CI 5.3)		
Torsional Buckling Stress	$f_{oz}=~82.2~{ m MPa}$		$\phi=~0.9$ pacity $M_s=~32.6~{ m km}$	$\mathbf{Nm} = f_{\mu I} \cdot \mathcal{Z}_{\nu} \cdot y_{I}^{1-0}$		
AS4600-2005, Eqn 3.3.3.2(12) Conditions: $(default) \rightarrow \frac{G J}{(A \tau_{c}^{2})} \cdot \left(1 + \frac{\pi^{2} E}{(G J J_{c})}\right)$ Flexural-Torsional Factor	$\beta = 0.556$		@ Clear	Calcs		
S			⊔ What's New -	Improved		
		connections, diagrams, and more! New year, stacks of new ClearCalcs updates! We're excited to kick off 2019 with a bang with a bevy of new and imminent updates including new calculation templates and features.				
		Log in now and have a look, or read below to find out more.				
		Envelope	diagrams			
		the shear, mo	r than ever to graphically discer ment, and deflection forces act th all diagrams undated to a full	ing		



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and we didn't charge them a cent extra.